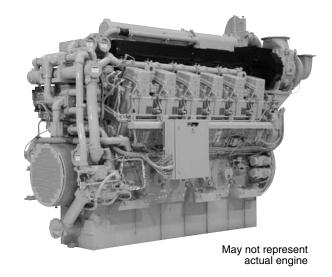
# **CATERPILLAR®**

C280-12 MARINE 3520 ekW AUXILIARY & DIESEL 3700 bkW (4962 bhp) ELECTRIC PROPULSION 50 Hz @ 1000 rpm



# **SPECIFICATIONS**

# V-12, 4-Stroke-Cycle-Diesel

| Emissions IMO II/EPA Tier 2 CBore 280 mm Stroke 300 mm Displacement 222 L (13,54 Rotation (from flywheel end) Countered Compression Ratio Aspiration Turbocharged-Aff Governor ELOW Idle Speed Speed Coil Change Interval* Serial Number Prefix Cooling System Keel or Heat ERefill Capacities | (11.0 in.)<br>(11.8 in.)<br>46 cu. in.)<br>clockwise<br>13:1<br>tercooled<br>Electronic<br>350 rpm<br>1000 rpm<br>750 hr |
|--|--|
| 3 ,  | xchanger   |
| Cooling System 1400-1575 L (370 Lube Oil System 910 L  | • ,  |
|  |  |

<sup>\*</sup>A new S•O•S<sup>sM</sup> analysis must be done to determine actual oil change intervals.

# STANDARD ENGINE EQUIPMENT

# **Air Inlet System**

Aftercooler — fresh water, corrosion resistant coated (air side); air inlet shutoff; breather — crankcase, top-mounted; turbocharger — engine oil lubricated

#### **Control System**

Single ADEM<sup>™</sup> A3 electronic control unit (ECU) with electronic unit injector fuel system, rigid wiring harness (10 amp, 24 volt power required to drive ECUs)

#### **Cooling System**

Engine coolant water drains

# **Exhaust System**

Dry, gas tight, exhaust manifold

#### **Fuel System**

Distillate fuel (requires viscosity ranging from 1.4 cSt to 20 cSt at 38° C), fuel transfer pump (mounted on left-hand side), duplex fuel filters, electronically controlled unit injectors

# **Lube Oil System**

Centrifugal oil filters with single shutoff, service side engine mounted on cylinder block inspection covers includes installed oil lines and single shutoff valve, filters centrifuge bypass oil from the main lubricating oil pump, can be serviced with the engine running, oil filler and dipstick valve, oil pressure regulating valves, crankcase explosion relief valves

#### General

Cat® yellow paint; gear-driven pumps: fuel, oil, jacket water, aftercooler/oil cooler water; service literature

Factory-designed systems built at Caterpillar ISO 9001:2000 certified facilities.

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# C280-12 MARINE AUXILIARY & **DIESEL ELECTRIC PROPULSION**

3520 ekW

# MARINE ENGINE PERFORMANCE

| C280-12   | DIESEL ENGINE T                         | ECHNICA | AL DATA                    | CA            | <b>TERP</b>  | LLAI        | R®         |
|---|---|---------|----------------------------|---------------|--------------|-------------|------------|
| Genset  | 50 Hz                                   |         | RATING:                    |               | Marine Aux - | Continuous  |            |
|   |   |         | CERTIFICATION              | N:            | IMO II/EPA N | MARINE TIEF | R II       |
| · (1)   | 1000                                    |         | TURBOCHARG                 | ER PART #:    |              |             | 189-4427   |
|   | 13:1                                    |         | FUEL TYPE:                 |               |              |             | Distillate |
| ( )   | 32                                      |         | RATED ALTITU               |               |              |             | 150        |
| ( )   | 90                                      |         | ASSUMED GEN                |               |              |             | 96         |
|   | EUI<br>DRY                              |         | ASSUMED GEN<br>MEAN PISTON |               |              | JK:         | 0.8<br>10  |
|   | 17300                                   |         | WEART ISTOR                | OI LLD (III/3 | ,,.          |             | 10         |
|   |   |         |                            |               |              |             |            |
| RATING  |   | NOTES   | LOAD                       | 110%          | 100%         | 75%         | 50%        |
| ENGINE POWER                                      |   | (2)     | bkW                        | 4070          | 3700         | 2775        | 1850       |
| GENERATOR POWER                                   |   | (2)     | ekW                        | 3907          | 3552         | 2664        | 1776       |
| BMEP  |   |         | kPa                        | 2204          | 2003         | 1503        | 1002       |
| ENGINE EFFICIENCY                                 | (ISO 3046/1)                            | (1)     | %                          | 42.1%         | 41.9%        | 40.4%       | 37.7%      |
| ENGINE EFFICIENCY                                 | (NOMINAL)                               | (1)     | %                          | 40.9%         | 40.6%        | 39.2%       | 36.5%      |
| ENGINE DATA                                       |   |         |                            |               |              |             |            |
| FUEL CONSUMPTION                                  | (ISO 3046/1)                            | (1)     | g/bkw-hr                   | 201.0         | 202.1        | 209.6       | 225.4      |
| FUEL CONSUMPTION                                  | (NOMINAL)                               | (1)     | g/bkw-hr                   | 204.9         | 206.0        | 213.7       | 229.8      |
| FUEL CONSUMPTION                                  | (90% CONFIDENCE)                        | (1)     | g/bkw-hr                   | 207.1         | 208.3        | 216.3       | 232.7      |
| AIR FLOW (@ 25°C, 101.3 kPaa)                     |   |         | Nm3/min                    | 410.0         | 386.0        | 313.4       | 236.8      |
| AIR MASS FLOW                                     |   |         | kg/hr                      | 27443         | 25833        | 20976       | 15852      |
| INLET MANIFOLD PRESSURE                           |   |         | kPa (abs)                  | 349.5         | 328.7        | 266.3       | 202.6      |
| INLET MANIFOLD TEMPERATURE                        |   |         | °C                         | 45.0          | 43.0         | 39.2        | 36.6       |
| EXHAUST STACK TEMPERATURE                         |   |         | °C                         | 377.0         | 373.5        | 371.8       | 365.7      |
| EXHAUST GAS FLOW (@ stack temp, 101.3             | 3 kPa)                                  |         | m3/min                     | 877.7         | 820.8        | 656.8       | 489.9      |
| EXHAUST GAS MASS FLOW                             |   |         | kg/hr                      | 28277         | 26596        | 21566       | 16271      |
| EMISSIONS "NOT TO EXCEED                          | DATA"                                   |         |                            |               |              |             |            |
| NOx (as NO) + THC (molecular weight of 13.        | 018)                                    |         | g/bkW-hr                   | 10.21         | 10.93        | 11.47       | 12.24      |
| NOx (as NO)                                       |   |         | g/bkW-hr                   | 9.29          | 10.02        | 10.26       | 10.55      |
| co  |   |         | g/bkW-hr                   | 0.67          | 0.75         | 0.85        | 1.31       |
| THC (molecular weight of 13.018)                  |   |         | g/bkW-hr                   | 0.92          | 0.91         | 1.21        | 1.69       |
| Particulates                                      |   |         | g/bkW-hr                   | 0.21          | 0.23         | 0.49        | 0.95       |
| EMISSIONS "NOMINAL DA                             | TA"                                     |         |                            |               |              |             |            |
| NOx (as NO) + THC (molecular weight of 13.        |   |         | g/bkW-hr                   | 8.79          | 9.41         | 9.85        | 10.48      |
| NOx (as NO)                                       | •                                       |         | g/bkW-hr                   | 8.08          | 8.72         | 8.92        | 9.18       |
| co  |   |         | g/bkW-hr                   | 0.52          | 0.58         | 0.65        | 1.01       |
| THC (molecular weight of 13.018)                  |   |         | g/bkW-hr                   | 0.71          | 0.70         | 0.93        | 1.30       |
| Particulates                                      |   |         | g/bkW-hr                   | 0.15          | 0.16         | 0.35        | 0.68       |
| ENERGY BALANCE DAT                                | · A · · · · · · · · · · · · · · · · · · |         |                            |               |              |             |            |
| FUEL INPUT ENERGY (LHV)                           | (NOMINAL)                               | (1)     | KW                         | 9963          | 9106         | 7077        | 5062       |
| HEAT REJ. TO JACKET WATER                         | (NOMINAL)                               | (3)     | KW                         | 807           | 760          | 640         | 505        |
| HEAT REJ. TO SACKET WATER HEAT REJ. TO ATMOSPHERE | (NOMINAL)                               | (4)     | KW                         | 199           | 182          | 142         | 101        |
| HEAT REJ. TO OIL COOLER                           | (NOMINAL)                               | (5)     | KW                         | 399           | 378          | 328         | 277        |
| HEAT REJ. TO EXH. (LHV to 25°C)                   | (NOMINAL)                               | (3)     | KW                         | 3150          | 2911         | 2379        | 1825       |
| HEAT REJ. TO EXH. (LHV to 177°C)                  | (NOMINAL)                               | (3)     | KW                         | 2394          | 2251         | 1856        | 1470       |
| HEAT REJ. TO AFTERCOOLER                          | (NOMINAL)                               | (6) (7) | KW                         | 1316          | 1155         | 801         | 500        |

CONSULT ALTITUDE CURVES FOR APPLICATIONS ABOVE MAXIMUM RATED ALTITUDE AND/OR TEMPERATURE CONSULT ALTITUDE CONSUMPTION ARE BASED ON 35 API, 16°C FUEL HAVING A LOWER HEATING VALUE OF 42.780 KJ/KG USED AT 29°C WITH A DENSITY OF 838.9 G/LITER

- 1) FUEL CONSUMPTION TOLERANCE. ISO 3046/1 IS 0, +5% OF FULL LOAD DATA. NOMINAL IS ±3 % OF FULL LOAD DATA 2) ENGINE POWER TOLERANCE IS ±3 % OF FULL LOAD DATA.
- 2) ENGINE POWER TOLERANCE IS ± 3 % OF FOLL LOAD DATA.

  3) HEAT REJECTION TO JACKET AND EXHAUST TOLERANCE IS ± 10% OF FULL LOAD DATA. (heat rate based on treated water 4) HEAT REJECTION TO ATMOSPHERE TOLERANCE IS ±50% OF FULL LOAD DATA. (heat rate based on treated water 1) HEAT REJECTION TO ATMOSPHERE TOLERANCE IS ±50% OF FULL LOAD DATA.
- 5) HEAT REJECTION TO LUBE OIL TOLERANCE IS ± 20% OF FULL LOAD DATA. (heat rate based on treated water 6) HEAT REJECTION TO AFTERCOOLER TOLERANCE IS ± 5% OF FULL LOAD DATA. (heat rate based on treated water
- 7) TOTAL AFTERCOOLER HEAT = AFTERCOOLER HEAT x ACHRF (heat rate based on treated water) 8) FUEL CONSUMPTION DATA IS WITHOUT SEA WATER PUMP.

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# C280-12 MARINE AUXILIARY & DIESEL ELECTRIC PROPULSION

3520 ekW

# MARINE ENGINE PERFORMANCE

C280-12

#### **DIESEL ENGINE TECHNICAL DATA**



| ALTI  | TUDE DE | RATION F | ACTORS |      | IMO II/EP | A MARIN | E TIER I |         |         |        |      |      |      |      |
|-------|---------|----------|--------|------|-----------|---------|----------|---------|---------|--------|------|------|------|------|
|       |         |          |        |      |           |         |          | _       |         |        |      |      |      |      |
|       | 50      | 0.94     | 0.91   | 0.88 | 0.86      | 0.83    | 0.81     | 0.78    | 0.76    | 0.74   | 0.71 | 0.69 | 0.67 | 0.65 |
|       | 45      | 0.95     | 0.93   | 0.90 | 0.87      | 0.85    | 0.82     | 0.80    | 0.77    | 0.75   | 0.73 | 0.70 | 0.68 | 0.66 |
| AIR   | 40      | 0.97     | 0.94   | 0.91 | 0.89      | 0.86    | 0.83     | 0.81    | 0.78    | 0.76   | 0.74 | 0.71 | 0.69 | 0.67 |
| TO    | 35      | 0.98     | 0.96   | 0.93 | 0.90      | 0.87    | 0.85     | 0.82    | 0.80    | 0.77   | 0.75 | 0.73 | 0.70 | 0.68 |
| TURBO | 30      | 1.00     | 0.97   | 0.94 | 0.92      | 0.89    | 0.86     | 0.84    | 0.81    | 0.79   | 0.76 | 0.74 | 0.71 | 0.69 |
|       | 25      | 1.00     | 0.99   | 0.96 | 0.93      | 0.90    | 0.88     | 0.85    | 0.82    | 0.80   | 0.77 | 0.75 | 0.73 | 0.70 |
| (°C)  | 20      | 1.00     | 1.00   | 0.98 | 0.95      | 0.92    | 0.89     | 0.86    | 0.84    | 0.81   | 0.79 | 0.76 | 0.74 | 0.72 |
|       | 15      | 1.00     | 1.00   | 0.99 | 0.96      | 0.93    | 0.91     | 0.88    | 0.85    | 0.83   | 0.80 | 0.78 | 0.75 | 0.73 |
|       | 10      | 1.00     | 1.00   | 1.00 | 0.98      | 0.95    | 0.92     | 0.89    | 0.87    | 0.84   | 0.82 | 0.79 | 0.77 | 0.74 |
|       |         | 0        | 250    | 500  | 750       | 1000    | 1250     | 1500    | 1750    | 2000   | 2250 | 2500 | 2750 | 3000 |
|       |         |          |        |      |           | ALTITUE | DE (METE | ERS ABO | OVE SEA | LEVEL) |      |      |      |      |

|       | AFTER | COOLER | HEAT RE | JECTION I | FACTORS | 3       |          |         |         |        |      |      |      |          |
|-------|-------|--------|---------|-----------|---------|---------|----------|---------|---------|--------|------|------|------|----------|
|       |       |        |         |           |         |         |          |         |         |        |      |      |      |          |
|       | 50    | 1.23   | 1.27    | 1.30      | 1.34    | 1.38    | 1.42     | 1.45    | 1.49    | 1.53   | 1.56 | 1.60 | 1.64 | 1.67     |
|       | 45    | 1.18   | 1.22    | 1.25      | 1.29    | 1.32    | 1.36     | 1.39    | 1.43    | 1.46   | 1.50 | 1.53 | 1.57 | 1.61     |
| AIR   | 40    | 1.13   | 1.17    | 1.20      | 1.23    | 1.27    | 1.30     | 1.34    | 1.37    | 1.40   | 1.44 | 1.47 | 1.50 | 1.54     |
| TO    | 35    | 1.08   | 1.12    | 1.15      | 1.18    | 1.21    | 1.24     | 1.28    | 1.31    | 1.34   | 1.37 | 1.41 | 1.44 | 1.47     |
| TURBO | 30    | 1.03   | 1.06    | 1.10      | 1.13    | 1.16    | 1.19     | 1.22    | 1.25    | 1.28   | 1.31 | 1.34 | 1.37 | 1.40     |
|       | 25    | 1.00   | 1.01    | 1.04      | 1.07    | 1.10    | 1.13     | 1.16    | 1.19    | 1.22   | 1.25 | 1.28 | 1.31 | 1.34     |
|       | 20    | 1.00   | 1.00    | 1.00      | 1.02    | 1.05    | 1.07     | 1.10    | 1.13    | 1.16   | 1.19 | 1.21 | 1.24 | 1.27     |
|       | 15    | 1.00   | 1.00    | 1.00      | 1.00    | 1.00    | 1.02     | 1.04    | 1.07    | 1.10   | 1.12 | 1.15 | 1.18 | 1.20     |
|       | 10    | 1.00   | 1.00    | 1.00      | 1.00    | 1.00    | 1.00     | 1.00    | 1.01    | 1.04   | 1.06 | 1.09 | 1.11 | 1.14     |
|       |       |        |         |           |         | ALTITUE | DE (METE | ERS ABO | OVE SEA | LEVEL) |      |      |      | <u> </u> |

#### FREE\_FIELD MECHANICAL NOISE

### SOUND PRESSURE LEVEL dB(A)

DISTANCE FROM 15M
THE ENGINE 7M
(M) 1M

| 109 96.2 102.2 101.7 102.3 101 | 3 99.3 98 95.6   |
|--------------------------------|------------------|
|                                |                  |
| 98 94.7 90.7 90.2 90.8 89.     | 8 87.8 86.5 84.1 |
| 92 79.2 85.2 84.7 85.3 84.     | 8 82.3 81 78.6   |

Octave Band (Hz)

#### FREE FIELD EXHAUST NOISE

#### SOUND PRESSURE dB(A)

DISTANCE FROM 15M
THE ENGINE 7M
(M) 1.5M

| 104         116.4         113.0         103.7         98.9         94.0         95.0         93.6           1         118         128.9         127.5         117.3         112.5         108.1         109.5         107.1 | 101.3 |        |      |      |       |       |       |     | 1 |
|---|-------|--------|------|------|-------|-------|-------|-----|---|
| 104 116.4 113.0 103.7 98.9 94.0 95.0 93.6   |       | 0 00.0 | 01.0 | 00.0 | 100.7 | 110.0 | 110.7 | 10- |   |
|   | 88.2  | 0 936  | 94.0 | 98.9 | 103.7 | 113 በ | 116.4 | 104 |   |
| 98 108.6 105.7 97.4 92.1 87.7 88.2 86.3   | 80.9  | 2 86.3 | 87.7 | 92.1 | 97.4  | 105.7 | 108.6 | 98  | ı |

Octave Band (Hz)

#### **TOTAL DERATION FACTORS:**

This table shows the deration required for various air inlet temperatures and altitudes. Use this information to help determine actual engine power for your site. The total deration factor includes deration due to altitude and ambient temperature, and air inlet manifold temperature deration.

#### AFTERCOOLER HEAT REJECTION FACTORS:

Aftercooler heat rejection is given for standard conditions of 25°C and 150 m altitude. To maintain a constant air inlet manifold temperature, as the air to turbo temperature goes up, so must the heat rejection. As altitude increases, the turbocharger must work harder to overcome the lower atmospheric pressure. This increases the amount of heat that must be removed from the inlet air by the aftercooler. Use the aftercooler heat rejection factor to adjust for ambient and altitude conditions. Multiply this factor by the standard aftercooler heat rejection.

#### GENERATOR EFFICIENCY:

Generator power determined with an assumed generator effeciency of 96% [generator power = engine power x 0.96]. If the actual generator efficiency is less than 96% [and greater than 94.5%], the generator power [ekW] listed in the technical data can still be achieved. The BSFC values must be increased by a factor.

The factor is a percentage = 96% - actual generator efficiency.

#### SOUND DATA

Data determined by methods similar to ISO Standard DIS-8528-10. Accuracy Grade 3.

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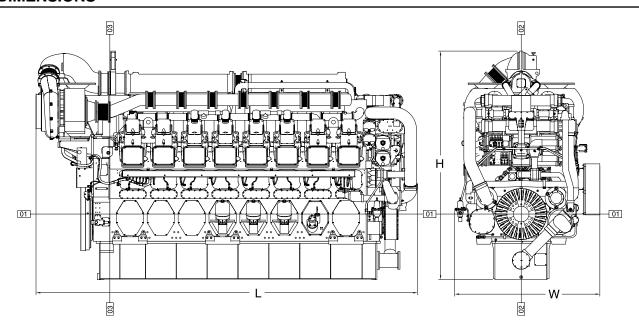
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# C280-12 MARINE AUXILIARY & DIESEL ELECTRIC PROPULSION

3520 ekW

# **DIMENSIONS**



| Engine Dimensions             |           |           |  |  |  |  |  |  |
|-------------------------------|-----------|-----------|--|--|--|--|--|--|
| Length of Engine              | 4087 mm   | 161.0 in  |  |  |  |  |  |  |
| Length with Generator         | 8040 mm   | 316.5 in  |  |  |  |  |  |  |
| Width of Engine               | 2000 mm   | 78.7 in   |  |  |  |  |  |  |
| Height                        | 4085 mm   | 160.8 in  |  |  |  |  |  |  |
| Engine Weight – dry (approx.) | 25,980 kg | 57,276 lb |  |  |  |  |  |  |
| Generator Weight – (approx.)  | 14,970 kg | 33,000 lb |  |  |  |  |  |  |

# **RATING DEFINITIONS AND CONDITIONS**

**Continuous Power** — 8,000 hrs./yr., for applications with load factors less than or equal to 100%. 10% overload available.

**Ratings** are based on SAE J1995/ISO3046 standard conditions of 100 kPa (29.61 in. Hg), 25°C (77°F), and 30% relative humidity at the stated charge air cooler water temperature. Ratings also meet classification society maximum temperature requirements of 45°C (113°F) air temperature to the turbocharger and 32°C (90°F) seawater temperature without derate.

Additional ratings may be available for specific customer requirements. Consult your Cat representative for additional information.

**Fuel rates** are based on 35° API, 16°C (60°F) fuel used at 29°C (85°F) with a density of 838.9 g/liter (7.001 lbs/U.S. gal). Lower Heat Value (LHV) of 42 780 kJ/kg (18,390 Btu/lb). Tolerance is +5%. Includes all engine mounted pumps. BSFC without pumps is 3% less.

Marine Certification — Ratings are marine classification society approved by ABS, BV, CCS, DnV, GL, LRS, and RINA. These societies have also granted C280 factory line production approval which eliminates requirement for society surveyor witness test.

Performance data is calculated in accordance with tolerances and conditions stated in this specification sheet and is only intended for purposes of comparison with other manufacturers' engines. Actual engine performance may vary according to the particular application of the engine and operating conditions beyond Caterpillar's control.

Power produced at the flywheel will be within standard tolerances up to 49°C (120°F) combustion air temperature measured at the air cleaner inlet, and fuel temperature up to 52°C (125°F) measured at the fuel filter base. Power rated in accordance with NMMA procedure as crankshaft power. Reduce crankshaft power by 3% for propeller shaft power.

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TMI Reference No.: DM8411-03